

GENERAL

Model Piper
 Type Piper PA-28RT-201 Arrow IV
 Engine rating 200 HP
 Engine model & type Lycoming, IO-360-C1C6
 Propeller McCauley B3D36C424/74SA
 Approx. Fuel burn /hr 11 USG/HR
 Powersettings % 75/65/55 APPROX. 12,4/ 10,8/ 9,1 USG/HR

BEM 859 KG
 BEM-ARM 87,19 IN
 BEM-Moment 164.789 Lbs/IN

ADVISORY PERFORMANCE (ISA, MSL, Max load)

Min RWY T/O (ISA, MSL, PAVED, LEVEL, DRY, MTOM, Flap 25°) GROUND RUN 320 M.
 OVER 50' 488 M.
 Min RWY LDG (ISA, MSL, PAVED, LEVEL, DRY, MTOM, Flap 40°) GROUND RUN 190 M.
 OVER 50' 463 M.

Endurance (No reserve) 7:25
 Max range (No reserve) 900 NM

LIMITATIONS

MTOM (Normal / Utility) 1250 KG.
 MLM (Normal / Utility) 1250 KG.
 MAX weight in baggage compartment 90 KG.
 Aft CG limit 93 IN
 FWD CG limit 90 IN

Oil capacity 8 U.S. Quarts
 Min. Operational oil quantity 6 U.S. Quarts
 Fuel capacity (Usable) 72 USG

Service ceiling 15.000 ft.
 Absolute ceiling 17.000 ft.

Max Demonstrated XWC 17 Kts
 Max tailwind (not stated in POH) 0 Kts

SPEEDS (IAS)

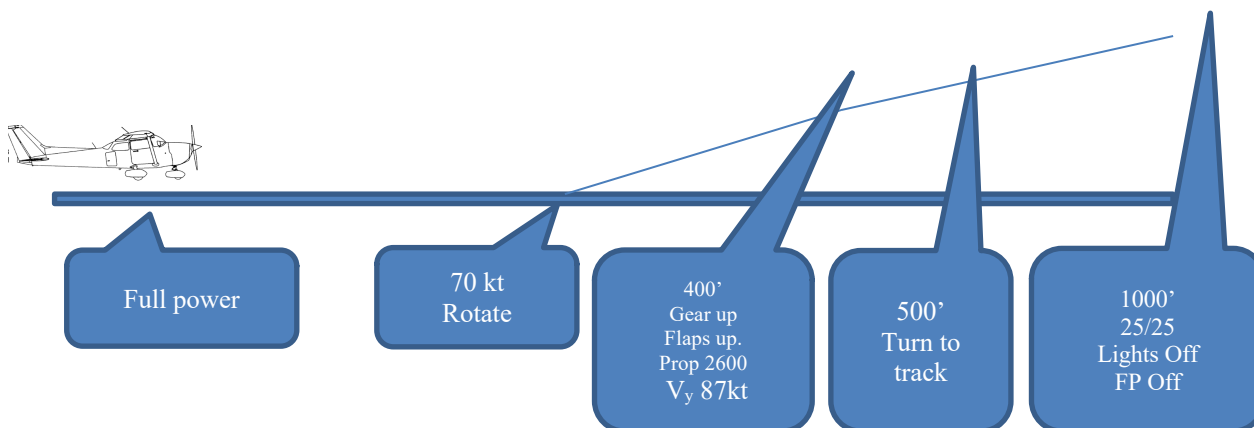
V_r Flap 25 (60 kts) 70 Kts
 V_{so} (Full Flaps & GEAR DOWN) 56 Kts
 V_{s1} (FLAPS UP & GEAR UP) 62 Kts
 V_x GEAR Down (70 Kts) 77 Kts
 V_y GEAR Down (76 Kts) 87 Kts
 V_{th} (flaps 40) 74 Kts
 V_{glide} 79 Kts
 V_a 2750 Lbs 121 Kts
 V_{no} 149 Kts
 V_{ne} 190 Kts
 V_{fe} 108 Kts
 V_{cc} (cruise climb) 104 Kts
 V_{cruise} 126 Kts
 V_{le} 130 Kts
 V (Gear Extension) 130 Kts
 V (Gear Retraction) 109 Kts
 Maximum window open speed 129 Kts

CERTIFICATIONS

Category: Normal
IFR **YES**
VFR YES
NIGHT VFR YES
ICING CONDITIONS NO
OPS where oxygen supply is necessary NO
Noise Certificate YES
Noise Certification STD ICAO Annex 16 Vol.1 Chapter 6

PAX Safety Briefing

Seat & Seatbelts OPERATION
Door / Emergency Exit(s) LOCATION & OPERATION
Fire Extinguisher LOCATION & OPERATION
Initial Emergency Checklist LOCATION & USE
Baggage STOWAGE
First Aid Kit LOCATION & CONTENT
Smoking NO SMOKING
Emergency procedure REVIEW PROCEDURE
PAX in front seat INFORM

1.1 Normal Takeoff

1.2 Short field

Flaps 25° (second notch)

Accelerate to 50 to 60 KIAS depending on aircraft weight.

Control Wheel back pressure to rotate to climb attitude

After breaking ground, accelerate to 55 to 65 KIAS depending on aircraft weight.

Gear (override engaged on aircraft equipped with backup gear extender) UP

Accelerate to best flaps up angle of climb speed - 77 KIAS, slowly retract the flaps and climb past the obstacle.

Accelerate to best flaps up rate of climb speed - 87 KIAS.

1.3 Soft field

SOFT FIELD

Flaps 25° (second notch)

Accelerate to 50 to 60 KIAS depending on aircraft weight.

Control Wheel back pressure to rotate to climb attitude

After breaking ground, accelerate to 55 to 65 KIAS depending on aircraft weight.

Gear (override engaged on aircraft equipped with backup gear extender) UP

Accelerate to best flaps up rate of climb speed - 87 KIAS.

Flaps retract slowly

1.4 Performance

Take-off performance

Altitude	Flap	TOR	TOD
0	0	518 m	685 m
2000	0	548 m	731 m
4000	0	594 m	807 m
0	25	320 m	488 m
2000	25	350 m	533 m
4000	25	396 m	580 m

Conditions: MTOW, ISA, no wind, dry paved runway

Climb performance

Altitude	ROC	Time	Distance
0	820 fpm	-	-
2000	740 fpm	2 min	4 NM
4000	650 fpm	5 min	8 NM
6000	560 fpm	9 min	15 NM
8000	470 fpm	12 min	20 NM
10000	380 fpm	17 min	28 NM

Cruise & PWR settings

STEP 1 Set RPM and PWR to 65%

STEP 2 Prelean, push LF and set LOP

STEP 3 Set PWR to 65% again

Altitude	Rpm rec.	PWR set to
Gnd – 4000	2300	65%
4000-6000 ft	2400	65%
6000-8000 ft	2500	65%
8000 –	2600	65%

Landing performance

Altitude	ACW	Ldg. Roll	Ldg. Dist
0	1248 kg (MTOW)	190 m	463 m
0	1062 kg*	160 m	433 m

Conditions: ISA, no wind, dry paved runway, 40 degree flaps, Vth = 74

1.5 Leaning procedure

Leaning LOP/ROP

1. Prelean.

Before entering LF mode, lean until EGT rise on all cyl

2. Select mode

Tap LF to start LF mode.

(Screen will show either ROP or LOP.

To toggle modes, press and hold Step and

LF together until display changes)

3. Start leaning

Start leaning slow - when 15 F change occurs the LF mode will activate

4. Lean of peak (normal operation)

When the last EGT peaks RICHEST will flash and the columns will turn upside down. Indicating you are now on the lean side on all cylinders.

Continue leaning to desired EGT or Fuel Flow (32 l for 65 %)

5. Rich of peak (only when demand for high power)

When first EGT peaks, LEANEST will

flash and flash the peaked EGT column. (If you press LF the peak EGT recorded will be displayed along with the FF spread)

Now enrichen to set 100F - 150 F ROP (of leanest cylinder) ensuring CHTs are below 380° F

1.6 Fuel

Fuel vs. payload (empty 859 kg)

FUEL LEVEL Dipstick markings = 10 L (Fuel hammer)	Volume/ Endurance per tank (useful fuel no reserve 40 l/h)	Payload (pilot, pax and bagg.) with equal tanks
0 = Less then	35 liter / 00:52 9,2 eller 18,4 Gal	352 kg
1	45 liter / 01:07 11,9 eller 23,8 Gal	338 kg
2	55 liter / 01:22 14,5 eller 29,1 Gal	323 kg
3	65 liter / 1:37 17,2 eller 34,3 Gal	309 kg
4	75 liter / 1:52 19,8 eller 39,6 Gal	295 kg
5	85 liter / 2:07 22,5 eller 44,9 Gal	280 kg
6 - TAP	95 liter / 2:22 25,1 eller 50,2 Gal	266 kg
7	105 liter / 2:37 27,7 eller 55,5 Gal	252 kg
8	115 liter / 2:52 30,4 eller 60,7 Gal	237 kg
9	125 liter / 3:07 33,0 eller 66,0 Gal	223 kg
10	135 liter / 3:22 35,7 eller 71,3 Gal	208 kg

1.7 Power settings

ISSUED: SEPTEMBER 14, 1979

REPORT: VB-1130

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POWER SETTING TABLE
Figure 5-21

LYCOMING MODEL IO-360-C SERIES, 200 HP ENGINE
— AS INSTALLED IN PA-28RT-201

Press. Alt. Feet	Std. Alt. Temp. °C	110 HP - 55% POWER RPM AND MAN. PRESS.		130 HP - 65% POWER RPM AND MAN. PRESS.		150 HP - 75% POWER RPM AND MAN. PRESS.	Press. Alt. Feet
		2200	2500	2200	2500	2500	
S.L.	15	25.0	22.2	27.1	23.5	26.0	S.L.
1000	13	24.6	21.8	26.8	23.3	25.8	1000
2000	11	24.1	21.5	26.4	23.1	25.7	2000
3000	9	23.7	21.2	26.1	23.0	25.6	3000
4000	7	23.3	20.9	25.7	22.9	25.5	4000
5000	5	22.8	20.5	F.T.	22.8	F.T.	5000
6000	3	22.4	20.2		22.7		6000
7000	1	21.9	19.8		22.5		7000
8000	-1	21.5	19.5		F.T.		8000
9000	-3	21.0	19.2				9000
10000	-5	F.T.	18.8				10000
11000	-7	—	18.5				11000
12000	-9	—	18.2				12000
13000	-11	—	17.8				13000
14000	-13	—	17.5				14000

To maintain constant power, correct manifold pressure approximately 0.16" Hg for each 10° F variation in inlet air temperature from standard altitude temperature. Add manifold pressure for air temperatures above standard; subtract for temperatures below standard.

Full throttle manifold pressure values may not be obtainable when atmospheric conditions are non-standard.

1.8 Autopilot

ROLL mode = master ON. Roll knob controls roll
HDG = Heading mode (see following)

Heading modes (set at HDG by default):

HDG = Heading bug (orange) on HSI (no wind corr.) or for use with GNSS setting on ADI

Modes with automatic wind correction +/- 15 degrees:

OMNI = VOR or GPS enroute

LOC = Localizer or GPS approach (normal) + GS

LOC BC = Localizer outbound (reverse) – no GS

Set Yellow arrow on HSI for desired track

Pitch modes requires ROLL or HDG engaged

PITCH mode = set pitch (not VS) on pitch wheel

ALT = Maintain present altitude (no altitude capture)

ALT + LOC = GS capture. Allow 20 sec straight and level flight with LOC and ALT engaged before capture